Ultra High Temperature, Miniature, SOI Sensors for Extreme Environments

Anthony D. Kurtz¹, Alexander A. Ned¹, and Alan H. Epstein²

¹Kulite Semiconductor Products, Inc. 1 Willow Tree Road Leonia, NJ 07605 U.S.A

ABSTRACT

The need for semiconductor pressure transducers that can operate at extremely high temperatures, in harsh environments that are corrosive, oxidizing, experiencing high vibration, is constantly on the rise across many industries.

This paper addresses the industry drive and discusses the latest developments of Silicon-On-Insulator (SOI) piezoresistive pressure sensors for extreme environments. The Design of the latest leadless miniature dynamic pressure transducer capable of operating reliably under extreme environmental conditions: 1) at temperatures in excess of 600°C (1100°F), 2) under accelerations/vibrations greater than 200g, and 3) in corrosive/oxidizing environments is described in detail. The performance of such leadless pressure transducers is presented and indicates that ruggedized, high frequency, acceleration compensated miniature. transducers with improved piezoresistive performance characteristics are now feasible for use in extremely harsh, high temperature environments. The Building Block approach used in construction of these transducers enables enormous flexibility in the design, thus enabling these sensors to be used across a wide range of industry applications.

INTRODUCTION

The sensor design for this latest generation of transducers utilizes the state-of-the-art MEMS technologies. In this design the front end of the sensor is intended to withstand both the high temperatures and the extreme environments throughout its lifetime, with an opportunity to locate the electronics (if needed) some distance ²MIT Room 31265 60 Vassar Street Cambridge, Mass 02139 USA

away. This approach enables the sensor to achieve high reliability under extreme operating There is no maintainability conditions. associated with this design approach since everything is enclosed and hermetically sealed within the transducer assembly. The transducer is designed to function without any maintenance. The modular building block approach used in this design together with the use of the SOI fabrication of piezoresistive sensors is intended to enhance transducer performance characteristics and to keep the cost of these transducers down.

The modern FADEC controls the starting, steady operation and transient conditions of a gas turbine and incorporates fault tolerant designs appropriate to a flight critical system. The ability of a FADEC to process the many sensor inputs from an engine, to apply sophisticated control laws and control a range of actuators simultaneously, enables many modern gas turbines to function economically and reliably. The next steps, which are predicted, to generate a significant improvement in efficiency of operation of a gas turbine have been identified as: 1) afterburner acoustic screech and rumble development and control, 2) the improvement in stability of compressors through the anticipation and suppression of surge and rotating stall, and 3) improvement in stability of the combustion process. The natural aerodynamic instabilities of turbomachines often limit their performance, but increased stability potentially leads to lighter more efficient compressors with fewer stages and shorter airfoil chords, reduced fan noise from lower tip speeds, faster engine acceleration as the surge constraints have been removed, and greater operating flexibility.

The piezoresistive approach adopted by Kulite is extremely well suited in terms of device performance, component utility, and systems integration to a variety of important applications on current and advanced airbreathing propulsion The extraordinary temperature systems. capability of these transducers, in the vicinity of 620°C (1150°F), combined with their high DC accuracy imply that they are applicable to a wide range of current and future pressure sensing applications. For example, compressor exit temperatures are now approaching 650°C (1200°F). So, in addition to afterburner screech development and control, which the transducer design is ideal for, these units can be used for measurement of compressor exit pressures. The DC and low frequency components would be used for conventional engine control logic, while the high frequency information is needed for stall avoidance and control and aeromechanical diagnostics. Of course, operational capability at compressor exit temperature implies that this technology is also well suited for main combustor measurements, both for development testing and active burner stabilization.

Piezoresistive transducers pursued herein are superior to alternative approaches for several reasons. First, unlike piezoelectric transducers, fragile controlled impedance wiring and bulky and expensive charge amplifiers are not required. Furthermore, the entire piezoresistive transducer and its leads are designed for the real engine, $>600^{\circ}C$ (1100°F) environment, while only the face of a piezoelectric can be exposed (the rest of the transducer and its leads must be cooled). While fiberoptic devices do not yet exist for flight engine environments, the ones proposed require relatively bulk, power consuming electro optics which must be provided with a cool, benign environment. On an engine, such an environment only exists with the fuel control, which is very dear real-estate. Thus, the new technology described herein offers the system level advantages of complete freedom from transducer cooling and thus more freedom for transducer placement, multifunction transducers, reduced transducer electronics, and reduced fuel control housing volume. Overall, this improves weight and reliability penalties the cost. associated with the additional control functionality desired for advanced engine concepts.

THE SILICON-ON-INSULATOR (SOI) SENSOR

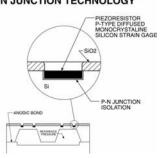
For the last 40 years, Kulite has supplied high performance pressure transducers to the aerospace industries for both research and development and for production applications. These transducers are based upon the piezoresistive silicon technology, which Kulite pioneered [1] and developed to its current high levels of performance and reliability. The latest evolution of sensors at Kulite, including the leadless, uses silicon on insulator (SOI) technology [2,3].

Piezoresistive silicon strain gauges are integrated within the silicon diaphragm structure but are electrically isolated from the silicon diaphragm as shown schematically in **Figure 1**.

The piezoresistors measure the stress in the silicon diaphragm, which is a direct function of the pressure of the media.

There are two core elements of the current generation of devices which will be considered in turn: the production of a suitable deflecting

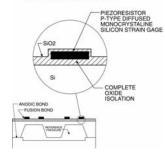
Silicon Integrated Sensors Diffused Pressure Sensor P-N JUNCTION TECHNOLOGY



- pn junction

- Temperature limited to 302°F (150°C)

Silicon on Silicon



- no pn junction
- Dielectrically isolated
- Temperature operation to 1100°F (593°C)

diaphragm to turn applied stress into displacement, and the addition of piezoresistive strain gauge elements to the diaphragm to record the displacement.

The latest evolution of the patented Silicon on Insulator (SOI) Technology enables the piezoresistive sensing elements to be dielectrically isolated from, while being molecularly attached to, a silicon diaphragm (**Figure 2**).

The process for fabricating the composite dielectrically isolated SOI sensor structure requires the use of two separate wafers. The first Pattern wafer is specifically selected to optimize the piezoresistive performance characteristics of the sensor chip, while the second Substrate wafer is specifically selected to optimize the micromachining capabilities of the sensing diaphragm. A layer of high quality thermally grown oxide is then grown on the surface of the substrate, while the piezoresistive patterns are introduced into the pattern wafer. The piezoresistive patterns are diffused to the highest level (solid solubility) in order to achieve the most stable, long-term electrical performance characteristics of the sensing network. Once the pattern and the substrate wafers are appropriately processed, the two wafers are fusion bonded together using a specifically developed and patented diffusion enhanced fusion technique [2]. The resulting molecular bond between the two wafers is as strong as silicon itself, and since

RESISTOR GAUGES IN THE THIN REGION SID 2 SID 2

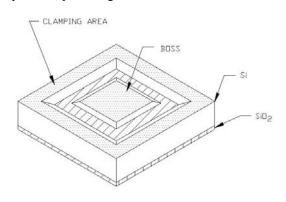
GAUGE SIDE ISOMETRIC OF SENSOR CHIP

mismatch between the two, thus resulting in very stable-accurate performance characteristics with temperature. The presence of dielectric isolation enables the sensor to function at very high temperatures without any leakage effects associated with the p-n junction type devices. Since the device is capable of operating at high temperatures, a high temperature metallization scheme is introduced to enable the device to interface with the header at these high temperatures as well.

The micromachining is performed using a combination of different wet (Isotropic and Anisotropic) chemical etch processes. The shape and performance characteristics of the micromachined sensing diaphragm are modeled using finite element analysis at the initial design stage. The composite silicon sensor is attached to a Pyrex pedestal, by an anodic bonding process, to form a pressure-sensing capsule as shown in **Figure 3**.

The pedestal material is selected to thermally match the physical characteristics of the silicon sensor.

The sensing circuit is electrically insulated from the metallic housing by virtue of the nonconductive pedestal in the pressure capsule. The isolation resistance and the dielectric strength are inherently very high. The reference pressure is accomplished between the diaphragm and pedestal, providing a true hermetic seal.



DIAPHRAGM SIDE ISOMETRIC OF SENSOR CHIP

Figure 2

both the sensing elements and the diaphragm are made from the same material, there is no thermal

Kulite Semiconductor Products, Inc. Presented at the IMAPS International HiTEC 2004 Conference Santa Fe, New Mexico, May 17-20, 2004

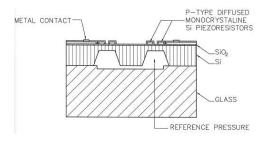


Figure 3 LEADLESS SENSOR DESIGN

Under extreme conditions of temperature and vibration, the ultrasonic agitation used to form the ball bonds causes abrasion to take place during the welding process and allows microscopic holes to develop in the platinum metallization through which, at high temperatures, the gold can migrate and form a gold-silicon eutectic which causes the leads to fail. In addition, the pressure media is in direct contact with the stress-sensing network, leadouts and interconnects, which at high temperatures and in the presence of aggressive chemicals, can fail. The key elements in the design of a ruggedized pressure sensor is the elimination of the gold bond wires and the protection of the sensing elements from corrosive environments at high temperatures, hence the reference to the new sensor capsule as the leadless design [4,5,6].

The leadless sensor capsule is comprised of two main components, the sensor chip and the cover wafer, which are eventually assembled to form the pressure capsule.

Leadless Sensor

Leadless Silicon on Silicon Sensor with Cover Glass Attached

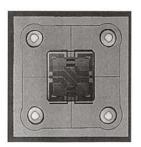


Figure 4

Figure 4 shows a photograph of the sensor chip with the four-piezoresistive gauges strategically positioned inside the sensing diaphragm region and connected in a Wheatstone bridge. The entire sensing network is P+ and there are

separations between the contact regions of the bridge. Metal is deposited to form ohmic contacts to the P+ regions located inside the large contact regions. There is also a rim of P+ material around the periphery of the sensor chip. When the cover wafer is assembled to the sensor chip, an hermetic seal is formed between the cover and this area of P+ material, thus protecting the stress sensing network and all the electrical interconnections from the harsh environmental conditions. The cover wafer is manufactured from a Pyrex glass to the same dimensions as the silicon wafer. Four holes are micromachined in the cover, one in each corner, which align with the metallized contact pad areas. A cavity is also created in the center of the cover wafer to allow the diaphragm to deflect freely when assembled. The sensor chip and the cover wafer are then assembled using an

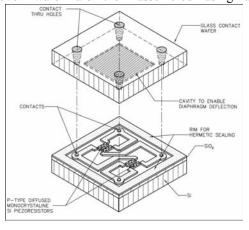


Figure 5

electrostatic bond.

Figure 5 shows a top isometric view of the components just prior to sealing. Once the two wafers have been bonded, only the metallized leadout pads are exposed while all the gauges and electrical interconnections on the sensing side of the silicon chip are sealed by the cover. Thus the active portion of the pressure sensor is hermetically isolated.

SENSOR IMPROVEMENT

Kulite is continuing to develop and optimize the sensor chip design and the associated packaging techniques. The latest generation of the optimized sensors includes: a) improvement of the overall performance characteristics of the sensor through finite element analysis and

Kulite Semiconductor Products, Inc. Presented at the IMAPS International HiTEC 2004 Conference Santa Fe, New Mexico, May 17-20, 2004 modeling, and b) increase of the temperature capability of the leadless sensor.

a) Finite Element Analysis modeling was performed to better understand the present designs and to fine-tune the new designs. The new designs were established to improve performance such as enabling larger, more linear, and more stable output characteristics for a specific sensor diaphragm thickness. Increasing the sensors output also created an opportunity to increase the diaphragm thickness for the respective design (in obtaining the same outputs), thus leading to improvements in the overall stability and repeatability of the sensors.

b) A significant effort is underway at Kulite to improve the metallization scheme on the sensing chip in order to enable device operation at ultra high temperatures 600°C (1100F). New metallurgical systems have demonstrated the capability of the metalized contacts on the sensing chips to withstand exposure to temperatures up to 650° C (1200°F) for many hours.

As part of our ongoing effort at Kulite for developing silicon pressure sensors for high temperature operation, we have investigated the behavior of sensor insulating layers at high temperatures. We have tested the dielectric isolation of SOI pressure sensors for temperatures up to 760°C (1400°F). Testing was performed by applying a DC voltage between the piezoresistive layer and the substrate, and by measuring the corresponding leakage current through the dielectric isolator. Testing was performed at wafer level, using a high temperature probe station. Isolation tests were done for voltages up to 100V without dielectric breakdown occurrence. Figure 6 shows leakage current values for applied voltages of 40V and 100V for temperatures ranging from 650°C (1200°F) - 760°C (1400°F) temperatures, and. Figure 7 shows the temperature dependence plot for leakage currents at 100V.

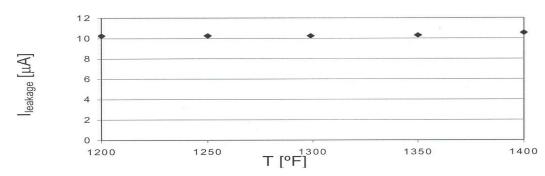
Figure 6

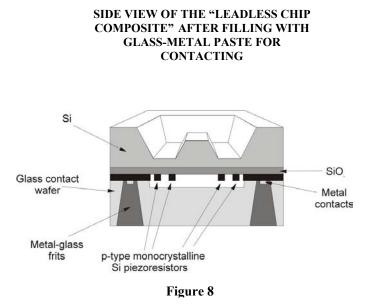
Temperature dependence of dielectric leakage currents for applied voltages of 40V and 100V

Temperature [°F]	I _{leakage} [μA] at 40V	I _{leakage} [μA] at 100V	
1200	4.10	10.25	
1250	4.10	10.26	
1300	4.09	10.25	
1350	4.15	10.30	
1400	4.30	10.55	

Figure 7

Temperature dependence of dielectric leakage current for 100 V applied voltage

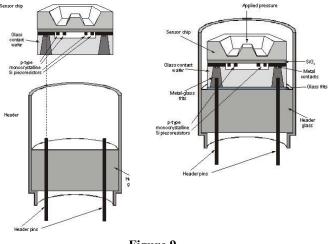




THE LEADLESS PACKAGING

To avoid the use of gold ball bonds and fine gold wires, a high temperature metal frit is used to provide the electrical connection between the sensing chip and a specially designed header. The frit is a mixture of high conductivity metal powders in appropriate physical form and glass, and is used to fill the holes in the cover wafer after it is attached to the sensor chip (**Figure 8**).

The specially designed header contains a group of four hermetically sealed pins protruding from its surface, which are spaced so as to fit the holes drilled in the cover wafer. The leadless sensor is bonded to the header at a high temperature using a non-conductive glass frit, during this process the metal frit in the cover wafer holes melts and



creates low resistance electrical connections between the header pins and the metal contact pads on the sensor chip. **Figure 9** shows the mounting process of a leadless chip onto a header and also a section of the pressure capsule mounted in the header.

Previous work [7] described transducers essentially made by these methods that were evaluated up to 580°C (1075°F). However, by varying the composition of the metal frit and its firing characteristics, and through continuous improvement in the metallization scheme used in the contact regions, significant improvements in high temperature reliability were obtained. The utilized leadless method allows the tip of the transducer to operate in excess of 600°C (1100°F). Once the chip is mounted onto the header, and the interconnections are established, only the non-active side of the diaphragm is exposed to the pressure medium. The small ball bonded gold leads have been eliminated and the entire sensor network and contact areas are hermetically sealed from the environment and the pressure media.

The hermetically sealed pressure-sensing capsule bonded to the header is the starting point for the assembly into a pressure transducer. Typically most transducers must be attached to a mounting surface, which is exposed to the pressure media, frequently by means of a threaded port. In addition, the header pins must be electrically connected to a high temperature cable assembly without the use of solder joints, which may fail at high temperatures. The high temperature cable assembly must also contain material which will provide electrical insulation between individual leads, while the interconnects between the header and the cable as well as the cable itself must be strong enough to withstand the mechanical stresses of handling. The package is completed using a building block approach and Figure 10 shows the assembly of a typical ultra high temperature leadless pressure transducer [8].

A sleeve is welded between the first header and a second header. A high temperature (HT) cable containing nickel wires is used to interconnect to the pins from the first header. The exposed leads from the first header are welded to the second header to ensure low resistance electrical connections between the leads of the HT cable

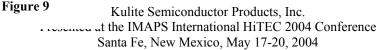
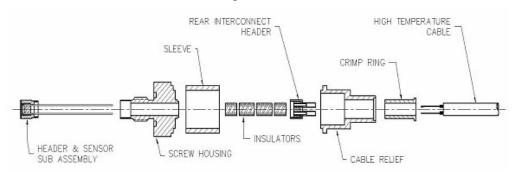


Figure 10

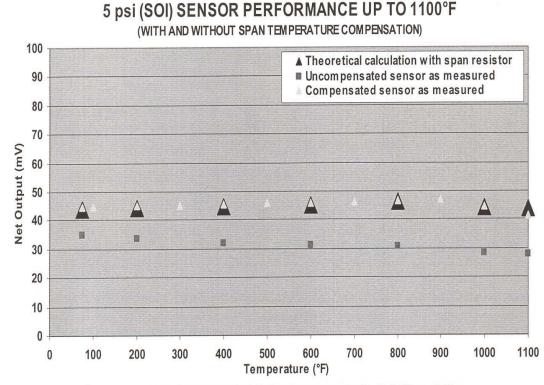


The header/ HT cable assembly is then inserted into a port and welded to the port. At the end of the port is a tubulation, which is crimped to retain the HT cable.

A cover sleeve is then assembled over the HT cable to give additional support and is welded to the rear of the cover, which in turn is welded to the port.

This design of assembly results in the transducer being totally hermetically sealed from any single internal metallized surface such as metal to silicon and metal to glass frit, header pins to header tubes, and header pins to HT cable wires are hermetically sealed from the atmosphere. In addition, the welding of the sleeve to the port and the addition of a cable relief and crimp ring greatly increases the structural integrity of the entire electrical interconnect system and reduces the chances of any damage in severe environments.

STATIC PERFORMANCE



Span compensated sensor calculated and measured with 10 V DC excitation Uncompensated sensor measured with 5 V DC excitation

Block Diagram of Shock Tube Set-Up

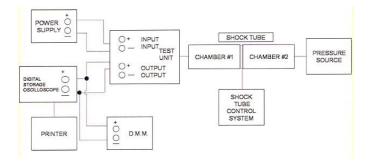


Figure 12

Test data from the latest generation of manufactured leadless transducers ranging from 5 PSI up to 1000 PSI have confirmed the original results and, in fact, has demonstrated operability of these sensors up to and above 600°C (1100°F). The latest sensors reconfirmed the previously reported data where devices were have excellent shown to performance characteristics. The new leadless SOI sensors were tested up to 600°C (1100°F). The data even for the low 5 PSI pressure units (Figure 11) was extremely stable and repeatable.

The 5 PSI units, compensated with a single span resistor exhibited excellent span and zero shifts over the entire temperature range from room temperature to 600°C (1100°F). All units tested exhibited only minor changes in performance characteristics after repeated exposure to high temperatures.

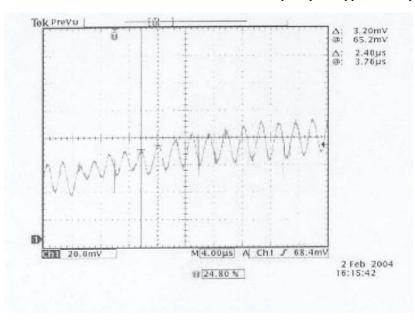
DYNAMIC PERFORMANCE

The design of the high temperature sensor is such that it should have high frequency response characteristics similar to those of more familiar, low temperature capability Kulite sensors. To verify this experimentally, a Dynamic Response Time Testing was performed using a shock tube. A block diagram of a shock tube set-up is shown in **Figure 12**.

The shock tube is an apparatus where a pressure medium is separated into two chambers that are isolated by a Mylar diaphragm. The device under test is mounted to the end of chamber #1 with its pressure-sensing element exposed to the inside of the chamber. A specified pressure level is applied to chamber #2 and the Mylar diaphragm is ruptured using the shock tube control system. This produces a square wave pressure pulse that travels through chamber #1 and excites the pressure-sensing element of the device under test. The dynamic response of the device under test is captured on the digital storage oscilloscope.

To test the dynamic response time of the latest leadless sensor, a high temperature 50 PSI device was mounted to the end of chamber #1. A pressure level of 50 PSI was placed in chamber #2. The shock tube control system pierced the diaphragm, which sent a square wave pressure pulse to the device under test.

The dynamic response of the device was captured on the digital storage oscilloscope, which was recorded to be 2.4 microseconds with a natural frequency of approximately 417 KHz

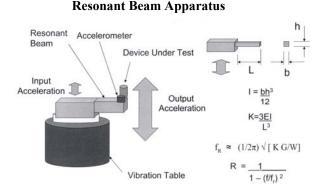


Natural Frequency = $1/t=\frac{1}{2}.4 \mu s=417 KHz$

8of 11

(Figure 13).

Additionally, in order to evaluate the robustness of its transducers, Kulite has developed and established its own testing and evaluation technology. Using standard vibration test apparatus, accelerations up to 50G can be achieved, in the 100 Hz to 3000 Hz range. In order to test components to significantly higher G levels, a resonant beam apparatus is used to achieve the high acceleration levels on standard vibration test equipment.

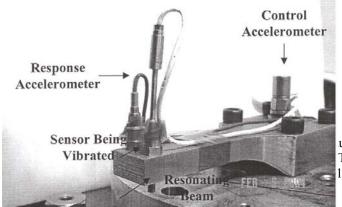


Natural Frequency = $1/t = \frac{1}{2}.4 \mu s = 417 Hz$

Figure 14

The beam is designed (Figure 14) to amplify the acceleration level around the beams resonant frequency with an amplification essentially being the Q of the system. The resonant frequency is chosen based on design requirements at the high acceleration level.

Typically, aircraft structures have the highest G requirements at about 1,000 Hz and the beam dimensions, such as length and cross section, are selected to achieve the frequency and amplification required. The beam is fastened to the vibration table (Figure 15), allowing the low



acceleration levels to act as the system input. The device under test, which will experience the amplified G level, is placed at the end of the beam. The amplification can be on the order of 10 to 250 times.

Two accelerometers were used on the beam to measure the vibration (acceleration levels). Accelerometer #1 was used to sense the shakerproduced vibration, while accelerometer #2 was used to measure the amplified vibration levels. A 50 PSI leadless transducer was mounted on the beam next to the measuring accelerometers (#2)and the beam was vibrated at its resonant frequency (at approximately 1100Hz) with a G of 2 in the shaker and a Q of 144.7. The transducer itself saw a G of 271 and the transducer was held at this vibration level for two (2) hours with no sign of degradation. By increasing the G level of the shaker to 15.3 and at the same resonant frequency, it resulted in a measured G level of 833 (Figure 16).

The transducer was held at that level for two (2) hours. Once again the units survived with absolutely no sign of degradation.

Control (Shaker G)	Beam (Resonating Frequency Hz)	System Q	Response (Output G)
1.98	1098	144.7	287
15.3	1098	54.4	833

Figure 16

ucts, Inc. TEC 2004 Conference 17-20, 2004

Figure 15

CONCLUSIONS

As part of an ongoing effort to increase both the temperature operability limit and the performance characteristics of the piezoresistive transducers, the latest generation of leadless sensors has been designed, fabricated and evaluated within Kulite with very encouraging results.

These sensors have been demonstrated to (1) operate up to and above 600°C (1100°F), (2) exhibit excellent static and dynamic performance characteristics (3) withstand very high G-level (acceleration) without any signs of degradation.

The SOI technology combined with the leadless packaging approach create an opportunity to push the silicon-based sensors even further. An effort to increase the temperature capability to even higher temperatures is presently underway. The results of this work will be subject of future technical papers.

REFERENCES

[1] A.D. Kurtz, "Development and Application of High Temperature Ultra Miniature Pressure Transducers", Fundamentals of Aerospace Instrumentation (ISA) Volume 3 1970.

[2] A.D. Kurtz, A.A. Ned, US Patent #5,286,671 issued to Kulite Semiconductor Products, Inc., Leonia, NJ 1994.

[3] A.A. Ned, A.D. Kurtz, "High Temperature Silicon on Insulated Silicon Pressure Sensors with Improved Performance Through Diffusion Enhanced Fusion (DEF) Bonding", International Instrumentation Symposium, Reno. NV, June 1998.

[4] A.D. Kurtz, A.A. Ned, US Patent #5,955,771 issued to Kulite Semiconductor Products, Inc., Leonia, NJ 1999.

[5] Dr. Anthony Kurtz, Alexander A. Ned, Robert Gardner, Scott Goodman, "Ruggedized High Temperature Piezoresistive Transducers", 45th International Instrumentation Society, Albuquerque, NM May 1999

[6] Dr. Anthony Kurtz, Alexander A. Ned, Dr. John W.H. Chivers, Dr. Alan Epstein, "Further Work on Ruggedized High Temperature Piezoresistive Transducers for Active Gas Turbine Engine Control", International Instrumentation Symposium, Seattle, WA, May 2000.

[7] Anthony D. Kurtz, Alexander A. Ned, Scott Goodman, Professor Alan H. Epstein, "Latest Ruggedized High Temperature Piezoresistive Transducers", NASA 2003 Propulsion Measurement Sensor Development Workshop, Huntsville, Alabama, May 13-15, 2003

[8] A.D. Kurtz, Scott Goodman, Robert Gardner, US Patent #6,363,792B1 issued to Kulite Semiconductor Products, Inc. Leonia, NJ 2002.